

$$F_r = \frac{U^2}{\gamma g}$$

It represents the ratios of the inertia force to the gravitational forces.

Euler's Number (Pressure coefficient):-

$$\frac{1}{C_p} = \frac{1}{E_u} = \frac{\text{Inertial force}}{\text{Pressure force}}$$

$$= \frac{\text{Mass} \times \text{Acceleration}}{\text{Pressure} \times \text{cross-sectional area}}$$

We know

$$\frac{e_2 (d\dot{q}_2/dt_2)}{e_1 (d\dot{q}_1/dt_1)} = \frac{\nabla_2 P_2}{\nabla_1 P_1}$$

$$\Rightarrow \frac{\nabla_2 P_2}{e_2 (d\dot{q}_2/dt_2)} = \frac{\nabla_1 P_1}{e_1 (d\dot{q}_1/dt_1)}$$

$$\Rightarrow \frac{P_2/l_2}{e_2 U_2/l_2 U_2^2} = \frac{P_1/l_1}{e_1 U_1/l_1 U_1}$$

$$\Rightarrow \frac{P_2 U_2^2}{e_2 U_2^2} = \frac{P_1}{e_1 U_1^2} = E_u$$

thus in flows where inertia and pressure forces predominate, the pressure coefficient must be the same for dynamic similarity to exist.

Mach Number  $\rightarrow$  The Perfect gas law and the velocity of sound is given by

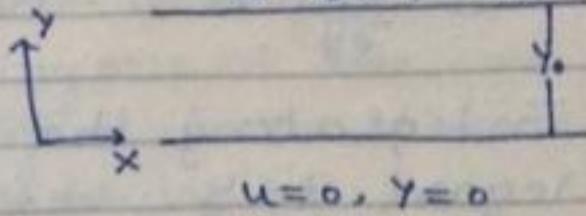
$$P = \rho R T \Rightarrow \frac{\gamma P}{\rho} = \gamma R T = a^2$$

where  $a$  is the speed of sound

11.0 Laminar flow between Parallel Plates:

S31

$$u = U, y = Y_0$$



Consider two-dimensional laminar flow of an incompressible fluid of constant viscosity between two parallel plates at a distance  $Y_0$ . The system does not regard to  $z$ -axis.

The equation of continuity

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0 \quad \text{--- (1)}$$

$$\rho \left( \frac{\partial u}{\partial t} + u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} \right) = - \frac{\partial p}{\partial x} + \rho g_x + \mu \left( \frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} \right) \quad \text{--- (2)}$$

$$\rho \left( \frac{\partial v}{\partial t} + u \frac{\partial v}{\partial x} + v \frac{\partial v}{\partial y} \right) = - \frac{\partial p}{\partial y} + \rho g_y + \mu \left( \frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial y^2} \right) \quad \text{--- (3)}$$

The flow between these plates is taken to be in the  $x$ -direction.

Then  $v = 0$ , or  $w = 0$  from the continuity equation, we have

$$\frac{\partial u}{\partial x} = 0 \Rightarrow u = u(y, t)$$

Hence the velocity dependent on  $y$  and  $t$ . The steady state  $u$  depends on  $y$  only  
 $u = u(y)$ ,  $v = 0 = w$  --- (4)

From (2) and (3) we have

$$0 = - \frac{\partial p}{\partial x} + \mu \frac{\partial^2 u}{\partial y^2} \quad \text{--- (5)}$$

$$0 = - \frac{\partial p}{\partial y}$$

From ⑥ we have

$$0 = -\frac{\partial P}{\partial y} \Rightarrow P = P(x)$$

Integrating the Equation ⑤ with regard to  $y$ , we have

$$\frac{dy}{dy^2} = \frac{1}{\mu} \frac{dP}{dx}$$

$$\Rightarrow \frac{dy}{dy} = \frac{1}{\mu} \frac{dP}{dx} y + A$$

$$\Rightarrow u = \frac{1}{2\mu} \frac{dP}{dx} y^2 + A y + B \quad \text{--- ②}$$

where  $A$  and  $B$  are integration constant

 Case I:- Plane Couette flow :- the plane Couette flow or simple shear flow between two parallel plates. the upper plate is moving in  $x$ -direction with a uniform velocity  $U$ . At the Lower plate

$$y=0, u=0 \Rightarrow B=0$$

$$\text{At the upper plate } y=Y_0, u=U \neq$$

$$\downarrow \text{we know } \mu \frac{du}{dy} = 0 \Rightarrow u(y) = A y + B$$

$$U = \frac{1}{2\mu} \frac{dP}{dx} \cdot 0 + A Y_0 + 0$$

$$A Y_0 = U$$

$$A = \frac{U}{Y_0}$$

$$\text{i.e. } \frac{dP}{dx} = 0$$

We have

$$u = \frac{U y}{Y_0}$$

It follows that the velocity profile induced in a fluid by moving one of the boundaries at constant velocity is linear across the gap between two boundaries.

### Case II - Generalized Plane Couette flow:-

In this case either of two surface is moving at constant velocity and there is also an external pressure gradient

$$\frac{dp}{dx} \neq 0$$

$$U = \frac{1}{2\mu} \frac{dp}{dx} y^2 + AY + B$$

$$Y = Y_0$$

$$U = \frac{1}{2\mu} \frac{dp}{dx} Y_0^2 + AY_0 + B$$

$$U = \frac{1}{2\mu} \frac{dp}{dx} Y_0^2 + AY_0 \quad [B=0]$$

$$A = - \left( \frac{U}{Y_0} + \frac{1}{2\mu} \frac{dp}{dx} Y_0 \right)$$

$$U(Y) = \frac{Uy}{Y_0} - \frac{Y_0^2}{2\mu} \frac{dp}{dx} \frac{y}{Y_0} \left( 1 - \frac{y}{Y_0} \right)$$

$$\Rightarrow \frac{U(Y)}{U} = \frac{y}{Y_0} + \left( - \frac{Y_0^2}{2\mu U} \frac{dp}{dx} \right) \frac{y}{Y_0} \left( 1 - \frac{y}{Y_0} \right)$$

$$\boxed{\frac{U(Y)}{U} = \frac{y}{Y_0} + P \frac{y}{Y_0} \left( 1 - \frac{y}{Y_0} \right)}$$

$$\text{where } P = - \frac{Y_0^2}{2\mu U} \frac{dp}{dx}$$

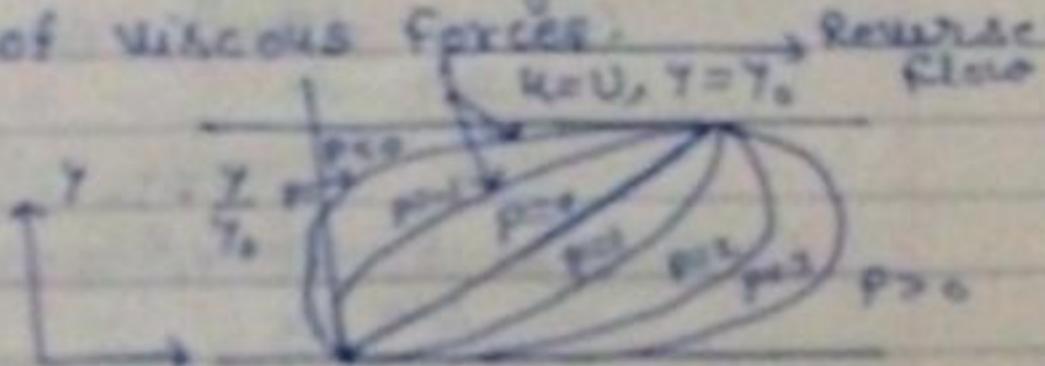
P is a dimensionless Pressure Parameter.

(ii)  $P > 0$  then  $\frac{dP}{dx} < 0$  in the direction of flow

the velocity is zero between the plates  
the pressure gradient will assist the viscosity induced motion to overcome the Shear Force at the lower plate.

(iii)  $P < 0$  then  $\frac{dP}{dx} > 0$  the Pressure gradient is  $\frac{dx}{dz}$  increasing in the direction of flow.

(iv)  $P = 0$  the fluid motion in the  $x$ -direction is entirely due to the action of viscous forces.



the average Velocity distribution  $V$

$$V = \frac{1}{L} \int_0^L u dy = \frac{1}{L} \int_0^L \left[ \frac{U}{\tau_0} + P \frac{\tau_0}{U} \left( 1 - \frac{y}{L} \right) \right] dy$$

$$\text{Let } z = \frac{y}{L} \text{ then } dz = \frac{1}{L} dy$$

$$V = U \int_0^1 [z + P(L-z)] dz$$

$$= U \int_0^1 (z + PL - Pz) dz$$

$$= U \left[ \frac{1}{2} + PL - \frac{P}{2} \right]$$

$$= U \left[ \frac{1}{2} + \frac{P}{2} \right]$$